



Engine Mini Pack

Engine Model:
RB211-535E4-37/19D

ESN: 31109

Time Since New: 72,701

Cycles Since New: 24,123

Time Since Last Shop Visit: 2479

Cycles Since Last Shop Visit: 794

Last Shop Visit: Ameco 2019

LLP Limiter: 1,039 @ Profile A&B
In accordance with SB72-AG875

Location: Spain

Available Immediately





SECTION 1

ENGINE RELEASE CERTIFICATE/FORM 8130



A J Walter Leasing Limited, Exchange House, 54-58 Athol Street, Douglas, IM1 1JD, Isle of Man, T: +44 7552 812215 E: engines@ajw-leasing.com W: www.ajw-aviation.com

1. Approving Competent Authority / Country <i>Autoridad Competente / País que otorga la Aprobación</i> AES A / KINGDOM OF SPAIN AES A / REINO DE ESPAÑA		2. AUTHORISED RELEASE CERTIFICATE EASA FORM 1 CERTIFICADO DE APTITUD PARA EL SERVICIO FORMULARIO EASA 1 EASA FORM 1 / FORMULARIO EASA 1			3. Form Tracking Number <i>Número de Seguimiento del Formulario</i> SAM202103243	
4. Organisation Name and Address <i>Nombre y Dirección de la Organización</i> SERVITEC S.A.		Ronda Santa Eulalia nº35 Nave 14. 08780 - Palrejà (Barcelona) -España-			5. Work Order / Contract / Invoice <i>Orden de Trabajo / Contrato / Factura</i> ISY-19022021-4R1	
6. Item <i>Elemento</i>	7. Description <i>Descripción</i>	8. Part Number <i>Número de Componente</i>	9. QTY <i>Cantidad</i>	10. Serial N° <i>Número de Serie</i>	11. Status/Work <i>Situación/Trabajo</i>	
1	POWERPLANT RB211	RB211-535E4-37	1	31109	OVERHAULED ☐ <i>Inspeccionado</i> REPAIRED ☐ <i>Reparado</i> INSPECTED/TESTED ☒ <i>Inspeccionado/Ensayado</i> MODIFIED ☐ <i>Modificado</i>	
12. Remarks <i>Observaciones</i> REMOVED SERVICEABLE FROM EC-ISY ON 20 TH MARCH 2021 IAW TASK CARD 71-999-11-01 REV.20 TH SEP 2020 TEST 5 "PERFORMANCE TEST" AND TEST 13 "TAKE-OFF POWER TEST" PERFORMED ON 7 TH MARCH 2021 IAW AMM 71-00-00715-089-R03 AND 71-00-00800-801-R03 AND 71-00-00715-223-R03 REV.127 POWERPLANT PRESERVED FOR MORE THAN 3 MONTHS ON 13 TH MARCH 2021 IAW AMM 71-00-03-620-801-R00 REV. 127 LAST BSI INSPECTION ON 8 TH MARCH 2021 AT TH: 72700:37, TC: 24123 AS PER WO PVG-ISY-19022021-2 WITHIN LIMITS DAMAGES / ENGINE SERVICEABLE WITH REGULAR TIME INSPECTION NEXT LLP DUE: IPC SHAFT STG 1-6; PN UL30176 SN DN60727 NO MISSING PART LIST NO AD/SB EMBODIED AT THIS TIME						
13a. Certifies that the items identified above were manufactured in conformity to : <i>Certifica que los elementos indicados se fabricaron de conformidad con:</i> ☐ Approved design data and are in condition for safe operation <i>Datos de diseño aprobados y están en condiciones de operar con seguridad</i> ☐ Non-approved design data specified in block 12 <i>Datos de diseño no aprobados especificados en la casilla 12</i>				14a. ☒ PART-145.A.50 Release to Service <i>PARTE 145.A.50 Aptitud para el Servicio</i> ☒ Other regulation specified in block 12 <i>Otras normas especificadas en casilla 12</i> Certifies that unless otherwise specified in block 12, the work identified in block 11 and described in block 12, was accomplished in accordance with PART-145 and in respect to that work the items are considered ready for release to service. <i>Certifica que salvo por lo indicado en la casilla 12, el trabajo señalado en la casilla 11 y descrito en la casillas 12 se ha llevado acabo de conformidad con la Parte - 145 y con respecto a este trabajo la(s) pieza(s) se consideran aptas para el servicio.</i>		
13b. Authorised Signature / Firma Autorizada		13c. Approval / Authorisation Number <i>Numero de la Aprobación / Autorización</i>		14b. Authorised Signature / Firma Autorizada		14c. Certificate / Approval Ref. N° <i>N° de ref. del certificado / aprobación</i> ES.145.209
13d. Name / Nombre		13e. Date(dd/mmm/yyyy) / Fecha (dd/mmm/aaaa)		14d. Name / Nombre MIGUEL MARTIN		14e. Date(d/m/y) / Fecha (d/m/a) dd/mmm/yyyy 24/03/2021

EASA FORM 1- MF/CAO/145 Issue 3 / Formulario EASA 1 - MF/CAO/145 Edición 3

USER / INSTALLER RESPONSABILITIES:

- This certificate does not automatically constitute authority to install the item(s).
- Where the user/installer performs work in accordance with regulations of an airworthiness authority different than the airworthiness authority specified in block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts items from the airworthiness authority specified in block 1.
- Statements in blocks 13a and 14a do not constitute installation certification. In all cases the aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

RESPONSABILIDADES DEL USUARIO / INSTALADOR:

- Este certificado no autoriza automáticamente a instalar el elemento.
- Si el usuario o instalador ejecuta el trabajo con arreglo a la normativa nacional de una autoridad de aeronavegabilidad distinta de la especificada en la casilla 1, es esencial que se asegure de que la autoridad a la cual está sujeta acepta los elementos de la autoridad de aeronavegabilidad especificada en la casilla 1.
- Las declaraciones 13a y 14a no constituyen una certificación de instalación. En todo caso, el registro de mantenimiento de la aeronave debe contener una certificación de instalación expedida por el usuario o instalador en virtud de la normativa nacional antes de poder poner la aeronave en vuelo.



SECTION 2

LLP SHEETS



A J Walter Leasing Limited, Exchange House, 54-58 Athol Street, Douglas, IM1 1JD, Isle of Man,
T: +44 1685 708572 E: alun.roberts@ajw-leasing.com W: www.ajw-aviation.com

INSTALL DATE:	21/mar/19
TSN =	72701:00
CSN =	24123
Date:	22/03/2021

ETT (AT DELIVERY)	70222:00
ETC (AT DELIVERY)	23329

TSESV =	2479:00
CSESV =	794
DATE (ESV) =	30-ene.-19

Remaining cycles to next Limit: 1039

PRIVILEGE
style

L1P DESC.	PART NUMBER	SERIAL #	TOTAL H @ DLVR	TOTAL HOURS	CYC A @ DLVR	CYCLES A	CYCLES B	CYCLES C	CYCLES D	CYCLES E	CYCLES F	CYCLES G	TOTAL CYCLES	LIMIT A	LIMIT B	LIMIT C	LIMIT D	LIMIT E	LIMIT F	LIMIT G	REM. A	REM. B	REM. C	REM. D	REM. E	REM. F	REM. G
01 MODULE																											
SHAFT ROTOR L P.C.	UL27910	DJATH0003A	30639:00	33118:00	10.659	1.592	0	0	0	9720	0	141	11.453	27.250	14.230	17.200	17.200	17.200	14.230	14.230	9.988	5.216	6.304	6.304	6.304	5.216	5.216
PC ROTOR DISC ASS	UL21343	STDK54882	13435:00	15914:00	4.483	1.592	0	0	0	3544	0	141	5.277	22.600	22.600	22.600	22.600	22.600	22.600	22.600	17.323	17.323	17.323	17.323	17.323	17.323	17.323
02 MODULE																											
IPC SHAFT STG 1-6	UL30176	DN60727	64460:00	66939:00	21.209	1.592	0	0	0	20270	0	141	22.003	26.100	26.100	26.100	25.000	22.700	26.100	22.700	1.039	1.039	1.039	996	904	1.039	904
03 MODULE																											
IPC REAR STUB SHAFT	UL20016	DAVR1163A	39184:00	41663:00	13.250	1.592	0	0	0	12311	0	141	14.044	26.500	26.500	26.500	26.500	26.500	26.500	26.500	12.456	12.456	12.456	12.456	12.456	12.456	12.456
04 MODULE																											
HPC STG 1-2 DISC	UL11678	MW0185299	29408:00	31887:00	10.106	794	0	0	0	10106	0	0	10.900	18.030	12.600	15.700	15.700	15.700	15.700	12.600	5.630	3.934	4.902	4.902	4.902	4.902	3.934
HPC STG 3 DISC	UL24738	SETM25440	13435:00	15914:00	4.483	1.592	0	0	0	3544	0	141	5.277	15.000	15.000	15.000	15.000	15.000	15.000	15.000	9.723	9.723	9.723	9.723	9.723	9.723	9.723
HPC REAR SHAFT	UL18114	MW0250332	30639:00	33118:00	10.659	1.592	0	0	0	9720	0	141	11.453	19.630	15.710	19.200	18.200	15.000	17.310	15.000	5.133	4.108	5.020	4.759	3.922	4.526	3.922
HPT DISC	UL39767	LDRCC216901	30639:00	33118:00	10.659	1.592	0	0	0	9720	0	141	11.453	19.500	14.700	15.000	14.500	14.000	10.000	14.000	4.173	3.145	3.210	3.103	2.996	2.140	2.996
05 MODULE																											
IPT DISC	UL27940	LDREB11583	13435:00	15914:00	4.483	1.592	0	0	0	3544	0	141	5.277	26.500	26.500	26.500	26.500	26.500	26.500	26.500	21.223	21.223	21.223	21.223	21.223	21.223	21.223
IPT SHAFT	UL37571	BAVP393	68721:00	71200:00	22.759	1.592	0	0	0	21820	0	141	23.553	26.500	26.500	26.500	26.500	26.500	26.500	26.500	2.947	2.947	2.947	2.947	2.947	2.947	2.947
LPT SHAFT	UL37584	SAEF800	30639:00	33118:00	10.659	1.592	0	0	0	9720	0	141	11.453	25.200	16.270	16.000	16.000	16.000	16.000	16.270	8.080	5.217	5.130	5.130	5.130	5.130	5.217
LPT DISC STG 1	UL29028	EETM3901	69178:00	71657:00	22.950	1.592	0	0	0	22011	0	141	23.744	27.650	27.650	27.650	27.650	27.650	23.200	27.650	3.906	3.906	3.906	3.906	3.906	3.277	3.906
LPT DISC STG 2	UL29029	SETM25485	13435:00	15914:00	4.483	1.592	0	0	0	3544	0	141	5.277	16.700	13.300	16.700	14.400	14.400	16.700	13.300	10.820	8.617	10.820	9.330	9.330	10.820	8.617
LPT DISC STG 3	UL29031	SETY01345	70222:00	72701:00	23.329	1.592	0	0	0	22390	0	141	24.123	25.630	25.630	25.630	25.630	25.630	25.630	25.630	1.507	1.507	1.507	1.507	1.507	1.507	1.507

07 MODULE	PART NUMBER	SERIAL #	TOTAL H @ DLVR	TOTAL HOURS	CYC A @ DLVR	TOTAL CYCLES	TIME LIMIT	CYCLES LIMIT	REM HOURS	REM CYCLES
BUSH SLEEVE HOUSING CENTER	FW18125	KY1746	13435:00	15914:00	4.483	5.277	36000	12000	20086	6.723
HOUSING, SPHERICAL BEARING	FW62968	202299163105	0:00	2479:00	0	794	N/A	36000	N/A	35.206

FORMULAS BASED ON TIME LIMITS MANUAL 01-04-09

PRIVILEGE
BOEING 757-767-777 style
**DIRECCIÓN TÉCNICA
MANTENIMIENTO**

A. Wang
[Signature]
22 March 2021

RB211-535-E4-3739

ESN 31109

PRV-MM-305-B757 R2 15/06/20

FLIGHT PROFILE A

INSTALL DATE:	21/mar/19	ETT (AT DELIVERY)	70222:00
TSN =	72701:00	ETC (AT DELIVERY)	23329
CSN =	24123		
Date:	22/03/2021		

TSESV =	2479:00
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DATE (ESV) =	30-ene.-19

Remaining cycles to next Limit: 1039

PRIVILEGE
style

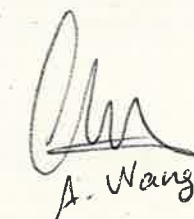
ANNULUS FILLER

PART NUMBER	SERIAL #	TOTAL H @ DLVR	TOTAL HOURS	CYC A @ DLVR	CYCLES A	CYCLES B	CYCLES C	CYCLES D	CYCLES E	CYCLES G	TOTAL CYCLES	LIMIT A	LIMIT B	LIMIT C	LIMIT D	LIMIT E	LIMIT G	REM. A	REM. B	REM. C	REM. D	REM. E	REM G
UL38852	200027HLAG66099	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG66932	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG66966	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG67154	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG67161	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG67175	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG67535	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG67876	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG67895	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG68036	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG68660	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG68697	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG69211	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG71773	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG71783	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG71927	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG71931	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG71936	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	200027HLAG71999	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	HLAG65740	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	HLAG65757	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620
UL38852	HLAG65768	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	12.000	12.000	12.000	12.000	12.000	12.000	9.620	9.620	9.620	9.620	9.620	9.620

FAN BLADES

PART NUMBER	SERIAL #	TOTAL H @ DLVR	TOTAL HOURS	CYC A @ DLVR	CYCLES A	CYCLES B	CYCLES C	CYCLES D	CYCLES E	CYCLES G	TOTAL CYCLES	LIMIT A	LIMIT B	LIMIT C	LIMIT D	LIMIT E	LIMIT G	REM. A	REM. B	REM. C	REM. D	REM. E	REM G
FW12372	GB81993	44248:00	46727:00	14.444	1.592	0	0	0	13505	141	15.238	24.580	17.600	23.000	23.000	23.000	17.600	8.358	5.984	7.821	7.821	7.821	5.984
FW12375	RGH17073	53020:00	55499:00	17.229	1.592	0	0	0	16290	141	18.023	24.580	17.600	23.000	23.000	23.000	12.000	5.290	3.787	4.950	4.950	4.950	2.582
FW12375	RGH17338	53020:00	55499:00	17.229	1.592	0	0	0	16290	141	18.023	24.580	17.600	23.000	23.000	23.000	12.000	5.290	3.787	4.950	4.950	4.950	2.582
FW12379	RGH22278	47859:00	50338:00	15.522	1.592	0	0	0	14583	141	16.316	24.580	17.600	23.000	23.000	23.000	12.000	7.114	5.094	6.657	6.657	6.657	3.473
FW12309	RRFASTRGH38241	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38256	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38266	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38275	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38494	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38511	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38517	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38531	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38539	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38544	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38547	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38555	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38562	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38566	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38572	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38581	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38594	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744
FW12309	RRFASTRGH38626	5420:00	7899:00	1.586	1.592	0	0	0	647	141	2.380	24.580	17.600	23.000	23.000	23.000	12.000	22.007	15.758	20.593	20.593	20.593	10.744

PRIVILEGE
BOEING 767-300-3739 style
DIRECCIÓN TÉCNICA
MANTENIMIENTO


A. Wang



SECTION 3

NON INCIDENT STATEMENT/NON GOVERNMENT USE STATEMENT



A J Walter Leasing Limited, Exchange House, 54-58 Athol Street, Douglas, IM1 1JD, Isle of Man, T: +44 7552 812215 E: engines@ajw-leasing.com W: www.ajw-aviation.com



Non-Incident / Non-Accident Statement:

Date: 22 March 2021

Engine Model: RB211-535E4-37/19D
MFR S/N: 31109

This Letter is to confirm that with regard to the above mentioned (**RB211-535E4-37/19D**) Engine bearing manufacturer serial number (**31109**), we hereby confirm that to the best of our knowledge the above mentioned Engine, together with its subparts and components, during its operation with Privilege Style from 19 March 2019 until 20 March 2021:

- Has not been involved in any incident, accident or failure as defined in Chapter 1 of ICAO Annex 13;
- Has not been subjected to fire, extreme stress or over temperature, immersed in salt water nor exceeded the Manufacturer's Operational Limitations;
- Has not been exposed to corrosive agents outside of those approved by the manufacture for the purposes of maintenance and/or preservation only; and
- Has not been acquired from a Military, Government entity or unapproved source, nor was subjected to any Military or Government operations.

Furthermore, during the time the above mentioned engine was operated by Privilege Style, we can confirm that this engine was not altered from its original approved Type Certificate or Type Design.

FINAL ENGINE TIMES: TSN 72701:00 – CSN 24123

Sincerely,

Signed: _____
Name: David Torres
Title: Technical Director

A handwritten signature in black ink, consisting of several overlapping loops and strokes, positioned over the signature line.

PRIVILEGE
style
BOEING 767-300ER
DIRECCIÓN TÉCNICA
MANTENIMIENTO



SECTION 4

LAST SHOP VISIT



AJ Walter Leasing Limited, Exchange House, 54-58 Athol Street, Douglas, IM1 1JD, Isle of Man, T: +44 7552 812215 E: engines@ajw-leasing.com W: www.ajw-aviation.com

发动机移交报告

ENGINE TRANSFER REPORT

R0

装机飞机型号(出厂) A/C OF INSTALL (OUT GOING)	发动机型号(出厂) ENG TYPE (OUT GOING)	发动机序号 ENGINE SN	用户 CUSTOMER	工作指令号 WORKORDER NO.
N/A	RB211-535E4-37/19D	31109	WAHOO	RP-000805
拆下日期 REMOVAL DATE	飞机/发位 AIRCRAFT/POSITION	机场 REMOVAL PLACE	发动机总时间 ENGINE TSN (FH)	发动机总循环 ENGINE CSN (CYC)
2017-08-08	P4-EAS/#2	UNK	70222	23329
拆发原因 REMOVAL REASON	LPT stage 2 blades rub 低涡二级转叶磨损			

发动机工作时限 ENGINE LIFE TIME LIMITS	详细信息见发动机工作时间报表 Details on the ENGINE TIME REPORT
装机单元体信息 INSTALLED MODULES	详细信息见发动机单元体报表 Details on the ENGINE MODULE AND UNITS REPORT
装机前/后需要 执行的工作信息 WORK TO BE PERFORMED BEFORE AND AFTER ENGINE INSTALLATION	安装发动机缺件单中所列的缺件。 Install the missing parts, please refer to "Engine missing parts list" 完成 AMM 中的测试及其他相关工作。 Accomplish the testing and related works required by AMM.
发动机油封信息 ENGINE PRESERVATION	发动机存储期限 3 个月到 24 个月(室内, 无空调; 存储日期 2019-01-30), 参考 EM E-211(535)-6RR Rev.139, 章节 72-00-00(存储), 请按照罗罗存储要求重新检查。验收时包封条件的任何破坏都会导致包封日期失效, 需要按照罗罗的说明重新检查。 Engine storage period is from 3 months to 24 months (inside, ambient-non air conditioned; stored on 2019-01-30) per EM E-211(535)-6RR Rev.139, chapter 72-00-00(storage) and please ensure the engine is storage and re-inspected iaw. RR requirements. Any interruption to the engine package condition at receipt may invalid the date with requires to be re-inspected as RR instruction.
车间工作描述 SHOP WORK DESCRIPTION	- 该发动机已参考EMP (RM1872 ISSUE 26) 及罗罗发动机手册E-211(535)-6RR的要求完成检查和修理(Level 2)工作。 - M32、M41、M51和M34进行了检查和修理 (Level 2), M31、M33和M61进行了目视检查(Level 1)。 - 故障件已按RB211-535E4发动机手册的标准完成修理或更换。 - 发动机试车台测试合格。 - 燃油系统及滑油系统已于2019年01月25日完成油封。 - The engine has been checked and repaired (Level 2) in accordance with the EMP (RM1872 ISSUE 26) and current effective Rolls-Royce Engine Manual E-211 (535)-6RR. - M32, M41, M51 and M34 were checked and repaired (Level 2), M31, M33 and M61 were serviceability (Level 1). - The defective parts have been repaired or replaced in accordance with the RB211-535E4 Engine Manual. - The engine successfully passed test-cell run. - Fuel system and oil system have been preserved on 2019-01-25.

此发动机已按相应发动机手册和合同维修工作要求完成了相关的车间维修工作, 可以移交。

The subject engine had been repaired and transferred per applicable engine manual and contract maintenance requirement for this shop visit.

填写日期:
ISSUE DATE: 2019-01-30

填写人:
ISSUED BY: 卓得 PRE 007355/0 李永 PRE 083759/0

第 1 页共 1 页
Page 1 of 1

Engine "Traveller": Hard copy, sent to the customer together with the engine;

Engine Logbook: Hard copy, sent to the customer at the time the engine shipment;

Scan copy: Electronic version, include whole set logbook files, sent to the customer by e-mail, earlier than the two others.

GROUP	Document Name	Engine "Traveller"	Engine Logbook	Scan copy	Note
Engine Level	Engine Release tag_FAA 8130	✓	✓	✓	
	Engine Release tag_EASA Form 1	✓	✓	✓	
	Statement For Aircraft/engine FAA AD Note Status Check	✓	✓	✓	
	FAA/EASA Airworthiness Directive Status	✓	✓	✓	
	List of implemented AD/SB	✓	✓	✓	
	Engine transfer report	✓	✓	✓	
	Engine outgoing missing parts list	✓	✓	✓	
	Engine test data sheet	✓	✓	✓	
	Engine module and unit report	✓	✓	✓	
	Engine time report	✓	✓	✓	
	LLP parts list	✓	✓	✓	
	Engine Accessory Report	✓	✓	✓	
	Engine outgoing borescope finding report	✓	✓	✓	
Module 32	Log card		✓	✓	
	Serialized parts record		✓	✓	
Module 41	Log card		✓	✓	
	Serialized parts record		✓	✓	
	Setting data		✓	✓	
Module 51	Log card		✓	✓	
	Serialized parts record		✓	✓	
	Setting data		✓	✓	
Module 34	Log card		✓	✓	
	Serialized parts record		✓	✓	
Non-module	Log card (Modifications applicable to non-modular engine sections)		✓	✓	

航空器 / 发动机 AD 指令状态检查声明

STATEMENT FOR AIRCRAFT/ENGINE AD NOTE STATUS CHECK

注册号 Aircraft Registration	制造厂 Make	序号 Serial No.	航空器/发动机型号 Aircraft/ Engine Model	工作完成日期 Date of Work Completed
	Rolls-Royce	31109	RB211-535E4-37	2019-01-25

声明:

Statement:

已对 ESN31109 发动机 FAA AD 状态进行了符合性检查, 符合要求。

The FAA AD status of the ESN31109 has been checked and it complies with AD requirements.

签署/日期:

Signed/Date:

张云朋

2019-01-25

批准人/日期:

Approved By/Date:

杨志

2019-01-25

Ameco Beijing

批准维修单位:

Maintenance Organization approval references:

FAA No.: XYJY995L

北京飞机维修工程有限公司

中华人民共和国北京首都机场, 邮政编码: 100621 邮政信箱: 563

Aircraft Maintenance and Engineering Corporation, Beijing.

(P. O.Box 563, Capital Airport, Beijing 100621, P. R. China)

ATTACHMENT OF RB211-535E4 ENGINE LOG BOOK

Aircraft Maintenance & Engineering Corporation

MODULE LOG CARD

MODULE 02

ENGINE TYPE. RB211 535E4 37/19D

ENGINE IDENTITY RECEIVED	SERIAL NUMBER	DESCRIPTION
M02626AA	DM4009	MODULE 02
ENGINE IDENTITY AT RELEASE	SERIAL NUMBER	DESCRIPTION
M02626AA	DM4009	MODULE 02
TIME SINCE NEW	CYCLE SINCE NEW	
64460	21209	

THE MODULE WAS CHECKED AND REPAIRED (LEVEL 2) AT AIRCRAFT MAINTENANCE & ENGINEERING CORPORATION, BEIJING (Ameco) ON 22 JAN. 2019 AND WAS FITTED TO ESN:31109

THE FOLLOWING AD/SERVICE BULLETINS WERE EMBODIED AT AMECO DATED 22 JAN. 2019.

Nil

Prepared
Signed: 卓爵 PRE 007355/0
Print name: 卓爵

Inspection
Stamp: 01210411
Date: 2019-01-29

RB211 535E4-37 SERIALISED PARTS RECORD MODULE 02	MODULE SERIAL No.	DM4009
	FITTED TO ENGINE No.	31109
	REMOVED FROM ENGINE No.	31109

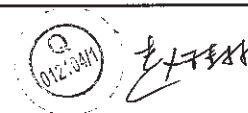
Code for rebuild

A	PREVIOUS PART SCRAPED	B	PART MODIFIED-REFITTED	C	PART RETAINED AT AMECO
D	PART NOT CHANGED	E	PART No. ERROR CORRECTED	F	SERIAL No. ERROR CORRECTED

DESCRIPTION	I.P.C ITEM No.	PART No.	SERIAL No.	CODE	TSN	Equivalent TFC (B)	DEC LIFE(B)	RFC(B)
Shaft rotor IPC	72-32-31 01-080	UL30176	DN60727	D	64460	24267	26100	1833
Housing BRG & IGV	72-32-11 01-140	Not disassembled		D	UNK	UNK	N/A	N/A
Bearing front LPC	72-32-11 03-420	Not disassembled		D	UNK	UNK	N/A	N/A
Bearing front IPC	72-32-11 03-920	Not disassembled		D	UNK	UNK	N/A	N/A
Case 1st rotor IPC	72-32-41 01-020	LK87995	AB62	D	UNK	UNK	N/A	N/A
Case Stage 2-6 IPC	72-32-41 01-050	UL31991	FT4	D	UNK	UNK	N/A	N/A
Front Stub Shaft, IPC	72-32-31 01-070	Not disassembled		D	UNK	UNK	N/A	N/A

NOTE:

TSN-Time Since New ; TFC(A/B)-Total Flight Cycles in Profile A or B
 DEC LIFE(A/B)-Declared Life in Profile A or B;RFC(A/B)-Remaining Flight Cycle in Profile A or B;
 The hours/cycles updated here are based on the LOGBOOK records.

Signed 	Approval Stamp
	

Date 2019-01-29

For Aircraft Maintenance & Engineering Corporation , Beijing

ATTACHMENT OF RB211-535E4 ENGINE LOG BOOK

Aircraft Maintenance & Engineering Corporation

MODULE LOG CARD

MODULE 04

ENGINE TYPE. RB211 535E4 37/19D

ENGINE IDENTITY RECEIVED	SERIAL NUMBER	DESCRIPTION
M04622AB	EM6253	MODULE 04
ENGINE IDENTITY AT RELEASE	SERIAL NUMBER	DESCRIPTION
M04622AB	EM6253	MODULE 04
TIME SINCE NEW	CYCLE SINCE NEW	
67583	23457	

THE MODULE WAS CHECKED AND REPAIRED (LEVEL 2) AT AIRCRAFT MAINTENANCE & ENGINEERING CORPORATION, BEIJING (Ameco) ON 22 JAN. 2019 AND WAS FITTED TO ESN:31109

THE FOLLOWING AD/SERVICE BULLETINS WERE EMBODIED AT AMECO DATED 22 JAN. 2019.

72-C903 72-J568 72-H660
 CAD 2003-B757-04R1/FAA AD 2004-19-04/CAA ADG-2003-0007
 72-AE359(CAD 2005-MULT-49/FAA AD 2007-14-07/CAA ADG-2005-0028R1)

Prepared
 Signed: 卓爵 PRE 007355/0
 Print name: 卓爵

Inspection
 Stamp: 
 Date: 2019-01-29

RB211 535E4-37 SERIALISED PARTS RECORD MODULE 04	MODULE SERIAL No.	EM6253
	FITTED TO ENGINE No.	31109
	REMOVED FROM ENGINE No.	31109

Code for rebuild

A	PREVIOUS PART SCRAPED	B	PART MODIFIED-REFITTED	C	PART RETAINED AT AMECO
D	PART NOT CHANGED	E	PART No. ERROR CORRECTED	F	SERIAL No. ERROR CORRECTED

DESCRIPTION	I.P.C ITEM No.	PART No.	SERIAL No.	CODE	TSN	Equivalent TFC (B)	DEC LIFE(B)	RFC(B)
Disc rotor HPT	72-41-51 01-350	UL39767	LDRCZ16901	D	30639	10956	14700	3744
Disc stage 1-2 HPC	72-41-31 01-170	UL11678	MW0185299	C	29408	8111	12600	4489
Disc stage 3 HPC	72-41-31 01-240	UL24738	SETM25440	D	13435	4483	15000	10517
Shaft rotor rear HPC	72-41-31 01-370	UL18114	MW0250332	D	30639	10967	15710	4743
Blade HPT	72-41-52 01-140	Not disassembled		D	UNK	UNK	N/A	N/A
Shaft of drive HPT	72-41-51 01-400	UL31964	CL10123	D	UNK	UNK	N/A	N/A
Case combustion	72-41-11 01-350	Not disassembled		D	UNK	UNK	N/A	N/A
Case comb.inner rear	72-41-12 01-050	Not disassembled		D	UNK	UNK	N/A	N/A
Liner comb.inner rear	72-41-13 01-650	Not disassembled		D	UNK	UNK	N/A	N/A
Liner comb.outer rear	72-41-14 01-940	Not disassembled		D	UNK	UNK	N/A	N/A
Liner comb. Front	72-41-13 02-250	Not disassembled		D	UNK	UNK	N/A	N/A
Case comb.inner FRT	72-41-12 03-500	Not disassembled		D	UNK	UNK	N/A	N/A
Case outer stg 1 HPC	72-41-41 01-070	KH31752	A004	B	UNK		N/A	N/A
Case outer stg 2 HPC	72-41-41 01-450	KH31753	A036	B	UNK		N/A	N/A
Case outer stg 3 HPC	72-41-41 02-450	KH31754	A039	B	UNK		N/A	N/A

RB211 535E4-37 SERIALISED PARTS RECORD MODULE 04	MODULE SERIAL No.	EM6253
	FITTED TO ENGINE No.	31109
	REMOVED FROM ENGINE No.	31109

Code for rebuild

A	PREVIOUS PART SCRAPED	B	PART MODIFIED-REFITTED	C	PART RETAINED AT AMECO
D	PART NOT CHANGED	E	PART No. ERROR CORRECTED	F	SERIAL No. ERROR CORRECTED

DESCRIPTION	I.P.C ITEM No.	PART No.	SERIAL No.	CODE	TSN	Equivalent TFC (B)	DEC LIFE(B)	RFC(B)
Cone locating centre	72-41-41 03-145	UL18118	N/A	D		UNK	N/A	N/A
Case outer stg 4 HPC	72-41-41 03-450	KH38315	A34	B		UNK	N/A	N/A
Case outer stg 5 HPC	72-41-41 04-450	UL18198	A05	D		UNK	N/A	N/A
Case outer stg 6 HPC	72-41-41 05-450	UL18200	31660	D		UNK	N/A	N/A

NOTE:

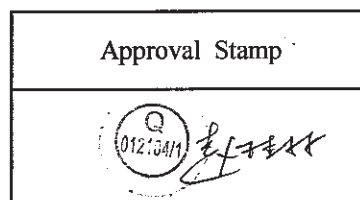
TSN-Time Since New ; TFC(A/B)-Total Flight Cycles in Profile A or B

DEC LIFE(A/B)-Declared Life in Profile A or B; RFC(A/B)-Remaining Flight Cycle in Profile A or B;

The hours/cycles updated here are based on the LOGBOOK records.

Signed  PRE
097355/0

Date 2019-01-29



For Aircraft Maintenance & Engineering Corporation , Beijing

MODULE 04 SETTING DATA

Ameco  **Beijing**

ENGINE TYPE RB211-535E4-37/19D	ENGINE No 31109	MODULE MK No M04622AB	MODULE SERIAL No EM6253
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DESCRIPTION OF DIMENSION	IDENT. LETTER	INFORMATION SOURCE	DIMENSION
HP TURBINE SETTING DIMENSION AT MODULE BUILD	EE1		7.654 mm
HP TURBINE SETTING AT ENGINE BUILD	EE2		7.629 mm

DATE: 2019-01-29 SIGNED: 卓育  PRE 087356/0

ATTACHMENT OF RB211-535E4 ENGINE LOG BOOK

Aircraft Maintenance & Engineering Corporation

MODULE LOG CARD

MODULE 05

ENGINE TYPE. RB211 535E4 37/19D

ENGINE IDENTITY RECEIVED	SERIAL NUMBER	DESCRIPTION
M05623AA	EM6328	MODULE 05
ENGINE IDENTITY AT RELEASE	SERIAL NUMBER	DESCRIPTION
M05623AA	EM6328	MODULE 05
TIME SINCE NEW	CYCLE SINCE NEW	
69351	22923	

THE MODULE WAS CHECKED AND REPAIRED (LEVEL 2) AT AIRCRAFT MAINTENANCE & ENGINEERING CORPORATION, BEIJING (Ameco) ON 22 JAN. 2019 AND WAS FITTED TO ESN:31109

THE FOLLOWING AD/SERVICE BULLETINS WERE EMBODIED AT AMECO DATED 22 JAN. 2019.

CAD 2003-B757-04R1/FAA AD 2004-19-04/CAA ADG-2003-0007

Prepared
Signed: 卓爵 PNE
C8785510

Print name: 卓爵

Inspection
Stamp: 01210411 卓爵

Date: 2019-01-29

RB211 535E4-37 SERIALISED PARTS RECORD MODULE 05	MODULE SERIAL No.	EM6328
	FITTED TO ENGINE No.	31109
	REMOVED FROM ENGINE No.	31109

Code for rebuild

A	PREVIOUS PART SCRAPED	B	PART MODIFIED-REFITTED	C	PART RETAINED AT AMECO
D	PART NOT CHANGED	E	PART No. ERROR CORRECTED	F	SERIAL No. ERROR CORRECTED

DESCRIPTION	I.P.C ITEM No.	PART No.	SERIAL No.	CODE	TSN	Equivalent TFC (B)	DEC LIFE(B)	RFC(B)
Disc rotor IPT	72-51-31 01-150	UL27940	LDREB11583	D	13,435	4,483	26,500	22017
Shaft IPT	72-51-33 01-185	UL37571	BAVP393	D	68,721	22,759	26,500	3741
Shaft LPT	72-51-63 01-200	UL37584	SAEF800	D	30,639	10,541	16,270	5729
Disc stage 1 LPT	72-51-61 01-300	UL29028	EETM3901	D	69,178	22,950	27,650	4700
Disc stage 2 LPT	72-51-61 01-350	UL29029	SETM25485	D	13,435	4,050	14,600	10208
Disc stage 3 LPT	72-51-61 01-470	UL29031	SETY01345	D	70,222	23,329	25,630	2301
Case exhaust LPT	72-51-71 02-350	Not disassembled		D	UNK	UNK	N/A	N/A
Bearing LPT	72-51-72 03-200	Not disassembled		D	UNK	UNK	N/A	N/A
Race inner IPT brg.	72-51-33 01-450	Not disassembled		D	UNK	UNK	NA	N/A
Race inner LPT brg.	72-51-61 01-550	Not disassembled		D	UNK	UNK	N/A	N/A
Case outer LP 2-3	72-51-51 01-060	UL12743	SSEAK1969	D	UNK	UNK	N/A	N/A
Case outer HP/IP brg.	72-51-21 02-065	Not disassembled		D	UNK	UNK	N/A	N/A
Bearing HPT	72-51-21 02-180	Not disassembled		D	UNK	UNK	N/A	N/A
Bearing IPT	72-51-21 02-530	Not disassembled		D	UNK	UNK	N/A	N/A
Blade rotor IPT	72-51-32 01-140	Not disassembled		D	UNK	UNK	N/A	N/A
Blade LPT stage 1	72-51-62 01-055	Not disassembled		D	UNK	UNK	N/A	N/A

RB211 535E4-37 SERIALISED PARTS RECORD MODULE 05	MODULE SERIAL No.	EM6328
	FITTED TO ENGINE No.	31109
	REMOVED FROM ENGINE No.	31109

Code for rebuild

A	PREVIOUS PART SCRAPED	B	PART MODIFIED-REFITTED	C	PART RETAINED AT AMECO
D	PART NOT CHANGED	E	PART No. ERROR CORRECTED	F	SERIAL No. ERROR CORRECTED

DESCRIPTION	I.P.C ITEM No.	PART No.	SERIAL No.	CODE	TSN	Equivalent TFC (B)	DEC LIFE(B)	RFC(B)
Blade LPT stage 2	72-51-62 01-090	UL20234	N/A	A	UNK	UNK	N/A	N/A
Blade LPT stage 3	72-51-62 01-140	Not disassembled		D	UNK	UNK	N/A	N/A

NOTE:

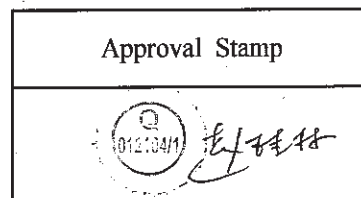
TSN-Time Since New ; TFC(A/B)-Total Flight Cycles in Profile A or B

DEC LIFE(A/B)-Declared Life in Profile A or B;RFC(A/B)-Remaining Flight Cycle in Profile A or B;

The hours/cycles updated here are based on the LOGBOOK records.

Signed  PRE
08735510

Date 2019-01-29



For Aircraft Maintenance & Engineering Corporation , Beijing

MODULE 05 SETTING DATA

Ameco  **Beijing**

ENGINE TYPE RB211-535E4-37/19D	ENGINE No 31109	MODULE MK No M05623AA	MODULE SERIAL No EM6328
-----------------------------------	--------------------	--------------------------	----------------------------

DESCRIPTION OF DIMENSION	IDENT. LETTER	INFORMATION SOURCE	DIMENSION
LP TURBINE SETTING DIMENSION (OUTER FLANGE TO LP SHAFT)	Q		5.827 in.
IP TURBINE SETTING DIMENSION (OUTER FLANGE TO LP SHAFT)	QW		5.781 in.
OUTER/INNER FLANGE DIMENSION	QA		3.801 in.
INNER FLANGE BEARING SUPPORT TO LP SHAFT	QB		2.026 in.
OVERALL WIDTH OF TAIL BEARING HOUSING	QC		
CLOCK READING IP COUPLING SETTING(BUILD STAND SETTING DIMENSION)	RG		
IP SHAFT DIMENSION	BV		34.566 mm
IP SETTING WITNESS	W		1.980 in.
IP NGV THROAT AREA			NOT APPLICABLE
LP NGV THROAT AREA			NOT APPLICABLE
IP TURBINE SETTING DIMENSION (MINI MODULE IN STAND)	RF		2.963 mm

DATE: 2019-04-29

SIGNED:  PEE 00735630

ATTACHMENT OF RB211-535E4 ENGINE LOG BOOK

Aircraft Maintenance & Engineering Corporation

MODULE LOG CARD

MODULE 07

ENGINE TYPE. RB211 535E4 37/19D

ENGINE IDENTITY RECEIVED	SERIAL NUMBER	DESCRIPTION
M07624AV	EM6309	MODULE 07

ENGINE IDENTITY AT RELEASE	SERIAL NUMBER	DESCRIPTION
M07624AV	EM6309	MODULE 07

TIME SINCE NEW	CYCLE SINCE NEW
70222	23329

THE MODULE WAS CHECKED AND REPAIRED (LEVEL 2) AT AIRCRAFT MAINTENANCE & ENGINEERING CORPORATION, BEIJING (Ameco) ON 22 JAN. 2019 AND WAS FITTED TO ESN: 31109

THE FOLLOWING AD/SERVICE BULLETINS WERE EMBODIED AT AMECO DATED 22 JAN. 2019.

Nil

Prepared
Signed: 卓爵 PRC 007355/0
Print name: 卓爵

Inspection
Stamp: (012/04/1) 卓爵
Date: 2019-01-29

RB211 535E4-37 SERIALISED PARTS RECORD MODULE 07	MODULE SERIAL No.	EM6309
	FITTED TO ENGINE No.	31109
	REMOVED FROM ENGINE No.	31109

Code for rebuild

A	PREVIOUS PART SCRAPPED	B	PART MODIFIED-REFITTED	C	PART RETAINED AT AMECO
D	PART NOT CHANGED	E	PART No. ERROR CORRECTED	F	SERIAL No. ERROR CORRECTED

DESCRIPTION	I.P.C ITEM No.	PART No.	SERIAL No.	CODE	TSN	TFC (B)		DEC LIFE(B)	RFC(B)
Bushings spherical	71-21-01 01-538	FW18125	KY1746	D	13435	4483		36000 Hrs	22565
								12000 CYCs	7517
Sleeve housing centre thrust link	71-21-01 01-536	FW62968	202299163105	C	0	0	0	36000	36000
Case OGV LPC	72-34-14 01-300	Not disassembled		D	UNK	UNK		N/A	N/A
OGV LPC	72-34-15 03-500	Not disassembled		D	UNK	UNK		N/A	N/A
Ring flanged location	72-34-11 02-020	Not disassembled		D	UNK	UNK		N/A	N/A
Diaphragm LPC case	72-34-11 02-300	Not disassembled		D	UNK	UNK		N/A	N/A

NOTE:

TSN-Time Since New ; TFC(A/B)-Total Flight Cycles in Profile A or B

DEC LIFE(A/B)-Declared Life in Profile A or B;RFC(A/B)-Remaining Flight Cycle in Profile A or B;

The hours/cycles updated here are based on the LOGBOOK records.

Signed  PRE 087355/0
Date 2019-01-29



For Aircraft Maintenance & Engineering Corporation , Beijing

RB211-535E4 MODIFICATION CERTIFICATE**MODIFICATIONS APPLICABLE TO NON-MODULAR ENGINE SECTIONS**

ENGINE TYPE RB211 535E4-37/19D	ENGINE NO. 31109	DIS. 2015 ISSUE 2
--	----------------------------	-----------------------------

Certified that the modifications listed below have been incorporated in this engine.

THE FOLLOWING MODIFICATIONS INCORPORATED AT AMECO DATED 22 Jan.2019

72-8672

71-AG051

Prepared
Signed:PRE
00735500

Print name: 卓爵

Inspection
Stamp:

Date:

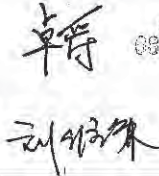
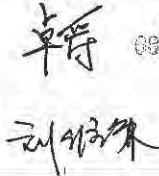
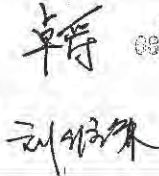
2019-01-29

For Aircraft Maintenance & Engineering Corporation, Beijing.

ENGINE MODULE AND UNITS REPORT

发动机单元体报表

Engine Type 发动机型号	Engine S/N 发动机序号	Engine TSN 发动机总时间	Engine CSN 发动机总循环	Removal Date 拆下日期	AC/Pos 飞机/发位	Work Order No. 工作指令号	Page 页号
RB211-535E4-37/19D	31109	70222	23329	2017/8/8	P4-EAS / #2	RP-000805	1/1
Module 单元体	Incoming 发动机进车间			Outgoing 发动机出车间			
Description 名称	Module P/N 单元体件号	Module S/N 单元体序号	Module TSN/CSN 单元体总时间/总循环	Module P/N 单元体件号	Module S/N 单元体序号	Module TSN/CSN 单元体总时间/总循环	
Module 31 LP Fan 31单元体 低压压气机	M01627AA	EM6256	72884/24525	M01627AA	EM6256	72884/24525	
Module 32 IP Compressor 32 单元体 中压压气机	M02626AA	DM4009	64460/21209	M02626AA	DM4009	64460/21209	
Module 33 Intermediate Case 33 单元体 中介机匣	M03622AA	DM4567	39184/13250	M03622AA	DM4567	39184/13250	
Module 41 HP System 41 单元体 高压单元体	M04622AB	EM6253	67583/23457	M04622AB	EM6253	67583/23457	
Module 51 IP/LP Turbine 51 单元体 中压/低压涡轮	M05623AA	EM6328	69351/22923	M05623AA	EM6328	69351/22923	
Module 61 HS Gearbox 61单元体 高速齿轮箱	M06627AB	EM6269	68132/23580	M06627AB	EM6269	68132/23580	
Module 34 Fan Case 34单元体 风扇机匣	M07624AV	EM6309	70222/23329	M07624AV	EM6309	70222/23329	
Unit 组件	Incoming 发动机进车间			Outgoing 发动机出车间			
Description 名称	Unit P/N 组件件号	Unit S/N 组件序号	TSN/CSN 总时间/总循环	Unit P/N 组件件号	Unit S/N 组件序号	TSN/CSN 总时间/总循环	
Exhaust Collector Common Nozzle 共用喷管收集器	JP55201-503	494	UNK/UNK	JP55201-503	494	UNK/UNK	
Cowling Side Panel LH 左部整流板	Incoming missing			Outgoing missing			
Cowling Side Panel RH 右部整流板	Incoming missing			Outgoing missing			
Distribution: 分发	Signature: 签名	PRE 087368/0	Date: 日期	Signature: 签名	PRE 013368/0	Date: 日期	

ENGINE TIME REPORT 发动机工作时间报告				Work Order No. 工作指令号		Page Of 第 1 页 共 1 页	
Engine Type RB211-535E4-37/19D 发动机型号		Engine S/N 31109 发动机序号		Engine TT 70222 总工作时间		Engine TFC 23329 总循环次数	
Removal Date 2017/8/8 发动机拆卸日期		Next Engine Removal At TT N/A 下次发动机拆下时总工作时间		Next Engine Removal At TFC N/A 下次发动机拆下时总循环数			
Module P/N 单元体件号	Module S/N 单元体序号	Module TT/TC 单元体总时间/总循环	MODULE TSR/CSR 单元体整修后时间/循环	Module Condition 单元体情况	Module RLT 单元体剩余时限	Module RFC 单元体剩余循环数	Remarks 备注
M01627AA	EM6256	72884/24525	13436/4483	Level 1	N/A	5,630	Profile B
M02626AA	DM4009	64460/21209	13436/4483	Level 2	N/A	1,833	Profile B
M03622AA	DM4567	39184/13250	13436/4483	Level 1	N/A	13,250	Profile B
M04622AB	EM6253	67583/23457	13436/4483	Level 2	N/A	3,744	Profile B
M05623AA	EM6328	69351/22923	13436/4483	Level 2	N/A	2,301	Profile B
M06627AB	EM6269	68132/23580	13436/4483	Level 1	N/A	N/A	Profile B
M07624AV	EM6309	70222/23329	13436/4483	Level 2			
Note: The hours/cycles here are based on customer documents. 注：小时数和循环数来源于客户文件。 Issued by 颁发  Date 日期 2019-01-29  							



SECTION 5

TEST CELL



J Walter Leasing Limited, Exchange House, 54-58 Athol Street, Douglas, IM1 1JD, Isle of Man,
T: +44 7552 812215 E: engines@ajw-leasing.com W: www.ajw-aviation.com

PRIVILEGE <i>style</i> Technical Department		Engineering Order		A/C. Reg.: ¹	EC-ISY
		EO number: ² ISY-04032021		Fleet: ³	B757
A/C Registr. affected: ⁴ EC-ISY		Originated by: ⁵ JCR		M/H Required: ⁶ -----	
Work Limit	Date: ⁷ -----	ACTT: ⁸ -----	ACTC: ⁹ -----		
Component	P/N: ¹⁰ RB211-535E4	S/N: ¹¹ ESN 31109	Pos: ¹² ENG#2		
Ordered Date: ¹³ 04/03/2021		Ordered to: ¹⁴ PART 145			
References: ¹⁵ AMM D633N143					
Enclosures: ¹⁶ N/A					
Title / Description: ¹⁷ ENG RH ESN 31109- ENGINE RUNS					
Work Specification: ¹⁸					
In accordance with re-delivery conditions do the following tests: - TEST No. 5 - PERFORMANCE TEST - on ENG RH ESN 31109. Ref. AMM TASK 71-00-00-715-089-R03 - TEST No.13 - TAKE-OFF POWER TEST - on ENG RH ESN 31109. Ref. AMM TASK 71-00-00-800-801-R03 and AMM TASK 71-00-00-715-223-R03.					
Material to be inst. Part Name		Part Number		Serial Number	
19		20		21	
Qty		22			
Findings / Remarks: (Cross out what applies)		☒ ²³ No ☐ ²⁴ Yes (if yes, note findings):			
NIL					
Certifies that the Work specified except as otherwise specified was carried out in accordance with PART-145 and in respect to that Work the aircraft/aircraft component is considered ready for release to service.				Date ²⁷	Time ²⁸
Name <u>DI NANNO</u> ³¹ Stamp  ³² Signature  ³³				Base / Station ²⁹	1LD
EASA PART M CAMO Approval reference: ES.MG.080		EASA PART-145 Organization Approval reference Number:		Technical Logbook seq. N° ³⁰ AAB 16361-4	
		N° <u>ES 145209</u> ³⁴			

IT	DESCRIPTION	Resp	To Fill in
1	Registration from aeroplane to which the Work defined as per item 18 is going to be carried out. If several aeroplanes are affected by the same EO, all the respective registrations must be included in item 4 "A/C Registr. Affected" and the Engineering Dept will be responsible of issuing the same EO for each of the registrations of the affected aeroplanes.	E	Before ordering EO
2	EO number. This number is automatically obtained from the computer application and is unique. Consist of 3 fields: XYYZZ-AA where: XX is the date of the month. YY is the month of the year. ZZ is the year AA is the sequence of the day (Ej: first EO is -01, second is -02)	E	Before ordering EO
3	Corresponding Fleet of the affected aeroplane. It will be shown as B757, B767, B777 or ERJ145.	E	Before ordering EO
4	This field must contain all the registrations of the aeroplanes affected by the EO.	E	Before ordering EO
5	Name of the person issuing the EO. Generally, only the Engineering Personnel will be allowed to issue the EO.	E	Before ordering EO
6	Man hours required to accomplish the tasks defined in the EO. This data is needed in order to optimally plan the work and for budget and economical control purposes.	E	Before ordering EO
7, 8 y 9	Fixed Deadline to accomplish the Work. This deadline must be indicated in one or several of the following fields:: Date (item 7), Aircraft Total Hours (item 8), Aircraft Total Cycles (item 9).	E	Before ordering EO
10, 11 y 12	- If the EO affects to an <u>Installed</u> Component, P/N and Position (Item 12) in the aircraft must be indicated. S/N will be indicated if this is expected to be found in the indicated Aircraft /Pos. If it is not found, the Certifying Technician must write down in the Findings (It. 20) box the S/N actually found. - If the EO affects to a <u>stored</u> Component, write down the P/N and, if applicable, S/N and indicate Stores in Item 14 (Ordered to). Therefore, Item 12 must not be filled in.	E	Before ordering EO
13	Indicate the Date of Issue of the EO by the Originator (whose name appears in item 5).	E	Before ordering EO
14	Indicate whom the completion of the EO is ordered to: Base, Station, Stores, Workshop.	E	Before ordering EO
15	Indicate the approved maintenance data reference on which the EO is based: EO, SB, SL, etc.	E	Before ordering EO
16	Indicate the documents attached to the EO and, if attaching parts of documents, indicate the applicable pages enclosed. Example, SB 32-310 (5 pages), MM 32-01-10 (3 pages)	E	Before ordering EO
17	Title of the EO. The title is created by the originator of the EO as to briefly describe the Work.	E	Before ordering EO
18	Detailed Specification of the tasks to be accomplished. Should the space in the box not be enough, the sentence "See N Pages Attached will be written down in this page" where N is the total number of attached pages required to describe the Work. These enclosed pages will keep the format defined for Work Specification, and in the upper right part for the Work specification It will be paged with Page 1 of N, Page 2 of N, etc.	E	Before ordering EO
19, 20, 21 y 22	Indicate Part name (it. 19), P/N (it. 20) and Quantity it. 22) of Material to be installed . When it is required, write down the S/N (item 21) of the component. This field informs Logistics about the material necessary to carry out EO (both components and tools, test equipments, etc).	E	Before ordering EO
23, 24, 25	- If the Certifying Technician does not find any defect during accomplishment of the Work, box NO (It. 23) will be crossed with an X. - Should the Certifying Technician find any defect during accomplishment of the Work, this must be described in the Findings/ Remarks (It. 25) box, crossing with an X the box YES (it. 24).	PC	At the conclusion of the Work
27, 28 y 29	Indicate the Date (It. 27), Z Time (It. 28) and Base or Station (It. 29) where the EO has been accomplished.	PC	At the conclusion of the Work
30	Logbook sequence number where the accomplishment of the EO was written down.	PC	At the conclusion of the Work
31, 32, 33 y 34	Name (It. 31), Stamp (It. 32) and Signature (It. 33) of the Certifying staff certifying the work performed. EASA Part-145 approval will be written down in item 34.	PC	At the conclusion of the Work

CAUTION: THIS ENGINE RUN PROFILE IS AN AID TO THE POWER PLANT TEST. FIRST, READ THE TEST INSTRUCTIONS. SECOND, COMPARE THE INSTRUCTIONS WITH THE RELATED PART OF THE PROFILE BEFORE YOU DO THE TEST. OBEY THE TEST INSTRUCTIONS TO PREVENT POWER PLANT DAMAGE.

TEST NO. 5	AIRPLANE NO.	EC-154
SHEET 2 OF 2	ENGINE SERIAL NO.	31109
	DATE	07/03/2021

TEST NO. 5 - PERFORMANCE TEST										ENGINE: L R				
Pressure Altitude		1060	Outside Air Temp.				13	Min Rated Thrust EPR				1.742		
Data Point	EPR Set	EPR Act.	N1		N2		N3		EGT		Oil Temp.	Oil Press.	Vib	
			Act.	Cor.	Act.	Cor.	Act.	Cor.	Act.	Cor.				
①	1.5	1.501	86.6	86.9	84.4	84.69	87.4	87.7	638	644.08	121	61	0.6	
②	1.58	1.568	90.4	91.1	87	87.3	89	89.3	671	677.31	129	58	0.8	
③	1.66	1.658	95.5	95.78	90.7	91.01	91.1	91.41	718	724.63	133	58	1.0	
④	1.74	1.736	98.9	99.25	93.5	93.83	92.6	92.916	750	756.85	136	58	0.9	

Test No. 5 - Performance Test
Figure 53771-00-00-990-880-R03 (Sheet 2 of 2)

EFFECTIVITY
IBE ALL

D633N143

ECN 8E391 BOEING PROPRIETARY - Copyright © Unpublished Work - See title page for details

71-00-00

Config 3
Page 598.120
Jan 20/2017

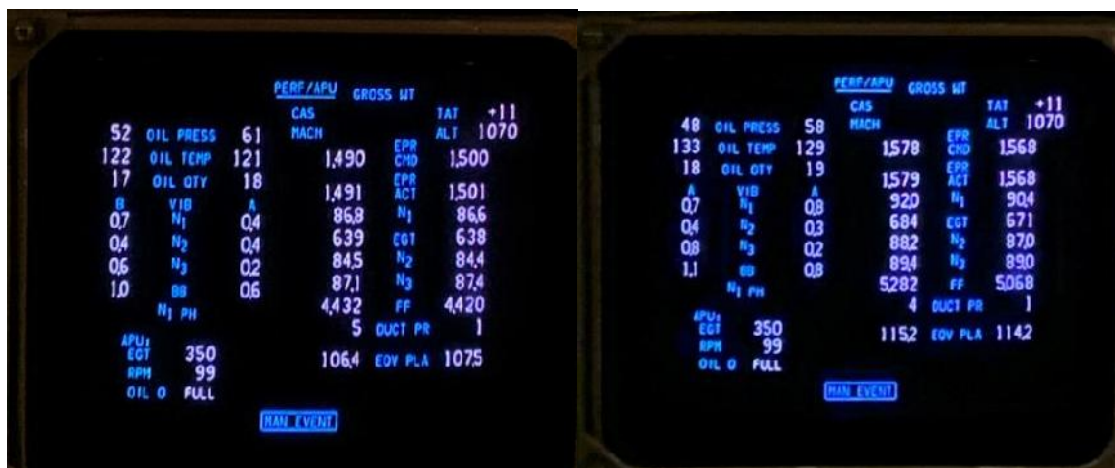
E39513 S0006127008_V1

PERFORMANCE TEST N.5 (AMM 71-00-00-715-089-R03)

RB211-535E4 ESN 31109 POS. RH

Lleida, Spain 7th March 2021

ALTITUDE		OAT		MINIMUM RATED THRUST EPR	
1400 ft		11 degrees		1.742	
EPR	1.5	1.58	1.66	1.74	
ACTUAL					
%N1	86.6	90.4	95.5	98.9	
%N2	84.4	87	90.7	93.4	
%N3	87.4	89	91.1	92.5	
EGT	638	671	718	750	
CORRECTED					
%N1	86.6	90.6	95.6	99.6	
%N2	84.5	87.6	91.6	93.6	
%N3	87.6	89.6	91.6	92.6	
EGT	647	682	728	763	
MAXIMUM					
%N1	87.6	91.8	96.2	101.4	
%N2	84.9	88.2	91.6	95.7	
%N3	89.1	91.1	93.1	95.4	
EGT	648	682	721	771	





SECTION 6

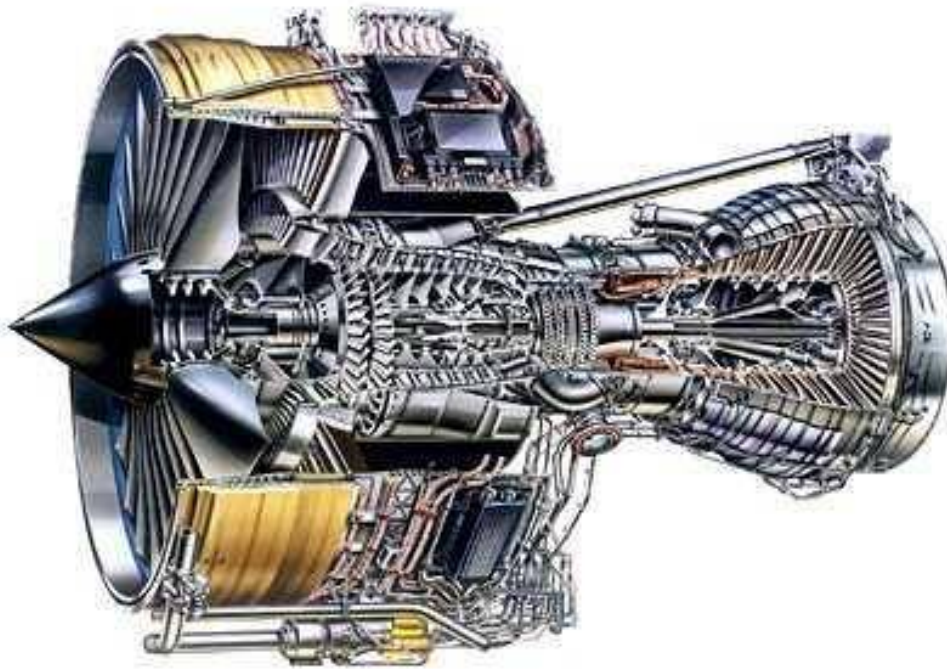
BORESCOPE REPORT



AJ Walter Leasing Limited, Exchange House, 54-58 Athol Street, Douglas, IM1 1JD, Isle of Man, T: +44 7552 812215 E: engines@ajw-leasing.com W: www.ajw-aviation.com

BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
This document must be always attached to its applicable Engineering Order or Approved Certificate of Release to Service Form		



BORESCOPE INSPECTION CONDITION			
SERVICEABLE	☒	UNSERVICEABLE	☐ (TICK AS APPLICABLE)
ADDITIONAL REQUIREMENTS:			
WITHIN LIMITS DAMAGES / ENGINE SERVICEABLE WITH REGULAR TIME INSPECTION INSPECTION PERFORMED AT ENGINE TOTAL HRS: 72700:37 TOTAL CYCLES: 24123			
CRS SIGN & STAMP	 CRS 05	DATE	08/03/21

BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
This document must be always attached to its applicable Engineering Order or Approved Certificate of Release to Service Form		
INSPECTION RESULTS		

IDENT PLATE



IPC STA 1 LE (ENGINE FRONT)

NIL DAMAGE / FINDINGS



BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
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IPC STA#2 TE (IP2 S)

NIL FINDINGS



IPC STA#3 LE (IP2S)

NIL FINDINGS



BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
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IPC STA #4 TE (IP4S)

NIL FINDINGS



IPC STA #5 LE (IP4S)



-2 BLADES FOUND WITH MISSING TIP. WITHIN LIMITS FOR IPC 4, LESS THAN 30% THRU CHROD WIDTH, NO RELATED CRACKS,NICKS OR TEARS AND SUBSEQUENT STAGES INSPECTED FOR DAMAGE. PERMITTED AS PER AMM 72-00-296-245-R02 PAR 1A 5F

BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
This document must be always attached to its applicable Engineering Order or Approved Certificate of Release to Service Form		



ONE BLADE WITH MISSING TIP FROM PREVIOUS REPAIR WITH SMOOTH CONTOUR. PERMITTED AS PER AMM 72-00-296-245-R02 PAR 1A 3 AND WITHIN LIMITS OF AMM 72-00-296-245-R02 PAR 1C 1, TOTAL NUMBER OF REPAIR BLADES IN BOTH HPC AND IPC IS NO MORE THAN 10.



BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
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IPC STA#6 TE (IP6)

NIL FINDINGS



HPC STA#1 TE

NIL FINDINGS



BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
This document must be always attached to its applicable Engineering Order or Approved Certificate of Release to Service Form		

HPC STA#2 LE (HP1S)

NIL FINDINGS



HPC STA#2 TE (HP2S)

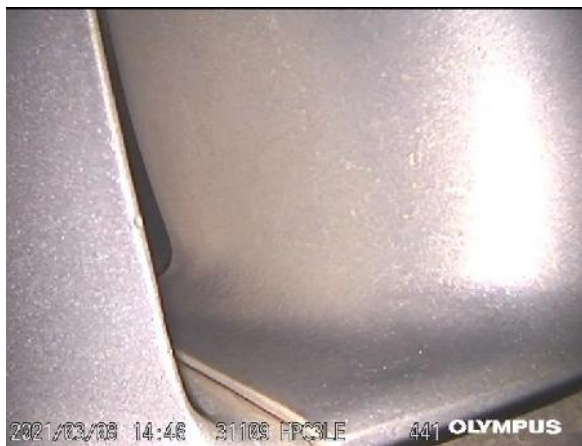
NIL FINDINGS



BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
This document must be always attached to its applicable Engineering Order or Approved Certificate of Release to Service Form		

HPC STA#3 LE (HP2S)



-ONE BLADE WITH A DENT WITHIN ZONE "D". NO RELATED CRACKS. WITHIN LIMITS AS PER AMM 72-00-846-155-R02 (4) (a) 4)



BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
This document must be always attached to its applicable Engineering Order or Approved Certificate of Release to Service Form		

HPC STA#3 TE (D)

NIL FINDINGS



HPC STA#4 LE (D)



BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
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-ONE BLADE BEND AT ZONE "C". WITHIN LIMITS /PERMITTED AS PER AMM 72-00-846-155-R02 (4) (a) 4)



HPC STA#5 TE (HP5S)



BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
This document must be always attached to its applicable Engineering Order or Approved Certificate of Release to Service Form		

-ONE BLADE BEND AT ZONE "D". WITHIN LIMITS /PERMITTED AS PER AMM 72-00-846-155-R02 (4) (a) 5)



HPC STA#6 LE

NIL FINDINGS



BORESCOPE INSPECTION REPORT RB211-535E4

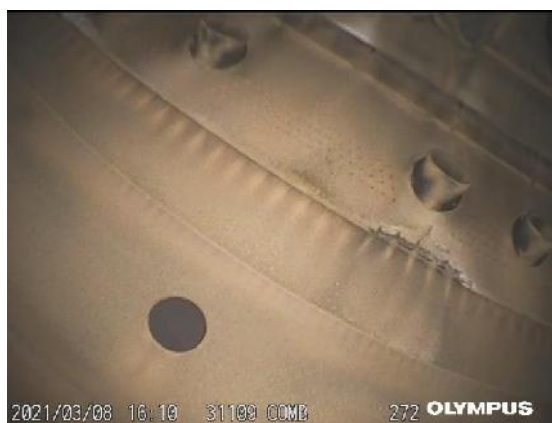
S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
This document must be always attached to its applicable Engineering Order or Approved Certificate of Release to Service Form		

COMBUSTION CHAMBER

-AT SEVERAL LOCATIONS, SIGNS OF DICOLORATIONS, MINOR BURNS AND EROSION. WITHIN LIMITS AS PER AMM 72-00-00-296-224-R02 (1) 2)



-AT SEVERAL LOCATIONS, MISSING MATERIAL WITHIN LIMITS AS PER AMM 72-00-00-296-224-R02 (1) 2)1



BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
This document must be always attached to its applicable Engineering Order or Approved Certificate of Release to Service Form		

-AT TWO LOCATIONS WITHIN FRONT INNER RING, METERING PANEL, MISSING MATERIAL WITH MINOR HOLING, NO MORE THAN 1.5SQ IN AND NO MORE THAN 10 LOCATIONS. PERMITTED AS PER AMM 72-00-00-296-224-R02-1 (C) 4) 1



BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
This document must be always attached to its applicable Engineering Order or Approved Certificate of Release to Service Form		

FUEL NOZZLES: NIL FINDINGS



HP NOZZLE GUIDE VANES

-SEVERAL LOCATIONS WITH MINOR BURNS AND EROSIONS, WITHOUT GOING INTO THE BASE MATERIAL.
PERMITTED AS PER AMM 72-00-00-200-004-R02 1) (E) 1)



BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
This document must be always attached to its applicable Engineering Order or Approved Certificate of Release to Service Form		

HP TURBINE LE

NIL FINDINGS



HP TURBINE TE

-SEVERAL BLADES WITH BURNS AT AREA B LOCATION XX WITHOUT MATERIAL MISSING. PERMITTED AS PER AMM 72-00-00296-171-R02-(1)c)(D)2)a)



BORESCOPE INSPECTION REPORT RB211-535E4

S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
AMM REF AND REV NUMBER	72-00-00/601 REV 127	
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-ONE BLADE WITH BURN AND MISSING MATERIAL AT AREA B LOCATION XX. MISSING MATERIAL CHECKED TO BE NOT MORE THAN 0.51MM AXIAL LENGTH. PERMITTED AS PER AMM 72-00-00296-171-R02-(1)c)(D)2)a)



-SEVERAL BLADES WITH CRACKS AT AREA B LOCATION XX. ALL CRACKS CHECKED TO BE NOT MORE THAN 3.18 MM AXIAL LENGTH. PERMITTED AS PER AMM 72-00-00296-171-R02-(1)c)(D)1)b)



BORESCOPE INSPECTION REPORT RB211-535E4

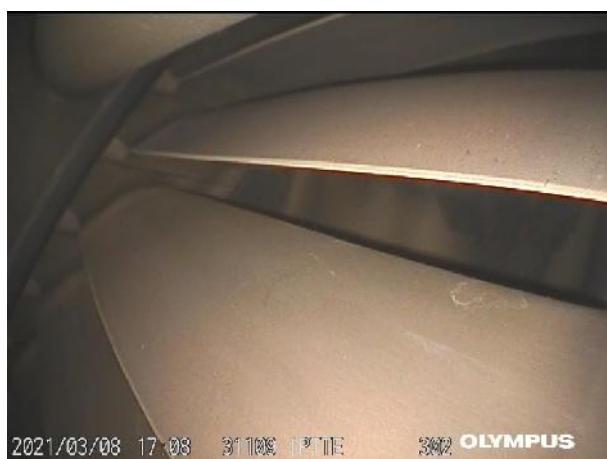
S/N:31109	CUSTOMER:PVG	E/O-W/O: PVG-ISY-19022021-2
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IP TURBINE LE

NIL FINDINGS



IP TURBINE TE



BORESCOPE INSPECTION REPORT RB211-535E4

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LPT STA#1 LE

NIL FINDINGS



LPT STA #1 TE

NIL FINDINGS



BORESCOPE INSPECTION REPORT RB211-535E4

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LPT STA#2 TE

NIL FINDINGS



LPT STA#3 LE



BORESCOPE INSPECTION REPORT RB211-535E4

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LP STA#3 TE

NIL FINDINGS





SECTION 7

AIRWORTHINESS DIRECTIVES (AD)



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AIRWORTHINESS DIRECTIVES STATUS REPORT

FAA BW 2021-04 and EASA BW 2021-04

Fecha / Date:	10/03/2021
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EASA/CAA	FAA	SUBJECT	AMEND	SB	ED	STAT	LAST ACC. DATE	LAST ACC. FH	LAST ACC. FC	INTERVAL			NEXT DUE			REMAINING			REMARKS
										DAYS	FH	FC	DATE	FH	FC	DAYS	FH	FC	
	75-16-08	to prevent damage in the intermediated and HPC that contributes to the frequency of high power surges		RB211-72-3326 RB211-72-3335 RB211-72-3482	22/08/1975	N/A													APPLICABLE TO ENGINE SN 10389 AND PRIOR
	75-16-09	To prevent engine loss of oil and possible inflight engine shutdown, modify the hs external gearbox oil tank filler cap and housing		RB211-72-3533	27/08/2005	N/A													APPLICABLE TO ENGINE SN 10479 AND PRIOR
	86-07-01	To prevent an uncontained outer combustion case burst		RB21-72-7775	16/05/2986	N/A													APPLICABLE TO PRE 72-8045 OUTER CASE ONLY. 72 8045 EMBODIED PREVIOUSLY
	87-25-04	To prevent low pressure (LP) fuel filter to high pressure (HP) pump tube fretage and cracking that can lead to fuel leaks and substantial fuel quantity loss	39-5766	RB211-73-8195 R02 RB211-73-8457 R02	08/12/1987	N/A													EMBODIED PREVIOUSLY DURING MANUFACTURE
	88-01-03	To minimize the potential for total thrust loss in both engines due to a lightning strike	39-5810	SB B757-71A0026 R04	16/02/1988	N/A													SB 71-8396 SUPERSEDED BY 71-8891, SB'S 71-8450, 71-8473, 71-8891 AND 75-8256 EMBODIED PREVIOUSLY
008-07-94	96-13-04 SUPERSEDED BY 98-07-07	NEW FLEXIBLE SECTIONS LP FUEL TUBE	39-9672	RB211-73-B048 R02	10/09/1996	SUP													SUPERSEDED BY 73-C297. EMBODIED PREVIOUSLY
005-05-96	97-02-12 SUPERSEDES 96-04-11R1	To prevent compressor stall and subsequent engine rundown on one or both engines	39-9897	RB211-73-B869 R03	04/04/1997	CW													EMBODIED PREVIOUSLY
007-10-97	98-07-07 SUPERSEDED BY 2014-24-08	INSPECTION LP FUEL TUBE FLEXIBLE SECTIONS	39-10426	RB211-73-C297 R04 RB211-73-B987 R05	29/04/1998	SUP													SB CANCELLED AT REV 4. SUPERSEDED BY FAA AD 2014-24-08
	98-25-12	ACCOUSTIC PANELS IN ENGINE INLET		RB211-71-9909 RB211-71-9958 RB211-71-B480R1	19/01/1999	N/A							N/A	N/A	N/A	N/A	N/A	N/A	N/A PER SN OF NOSE CORE
G-002-01-2000		LPC CRACKS IN LPC FAN BLADES ROOTS			27/09/2003	SUP													SEE AD 2008-22-24
CAA 001-07-86 R00 RT000		Ability of the engine to self recover is marginal, subsequent to bird/slush ingestion		RB211-75-8285 R00 RT000	N/A	PCW													EMBODIED PREVIOUSLY DURING MANUFACTURE
CAA 001-10-2000		Engine-HP compressor stage 4-6 rotor sha		RB211-72-D085 R01	N/A	N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	HPC DISC 4-6 INSTALLED NOT AFFECTED BY SB
CAA 002-05-2000		REVISED STOWAGE PROCEDURE & INSP OF N° 14 FUEL NOZZLE SPRAY		78-D011R4	N/A	N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	PHSE 5 COMBUSTOR IS NOT INSTALLED
EASA 002-05-87		HEALTH CHECK TO ESTABLISH HP COMPRESSOR SURGE MARGIN AND REVISED OPERATING PROCEDURES TO REDUCE THE POSSIBILITY OF COMPRESSOR SURGE ON TAKE-OFF			22/11/2012	N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	APPLICABLE TO BOEING 757 AIRCRAFT FITTED WITH RB211-535C ENGINES
CAA 004-08-2000R1 SUPERSEDED BY 005-04-20020		FW AND REAR ENGINE MOUNTS INSPECTION		71-5291R16		SUP				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	CANCELLED AND SUPERSEDED BY CAA AD 005-04-2002
EASA 004-08-86		ENGINE INDICATING-THERMOCOUPLE HARNESS-INTRODUCTION OF A REVISED VALUE OF TURBINE (TGT) TRIM BALLAST RESISTOR TO CODE W-MOD			22/11/2012	N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	APPLICABLE TO ENGINES PRIOR ESN 30104
004-12-99	2000-09-14	RB211-535 radial drive steady bearing	39-11723	RB211-72-C925 R04 RT000	11/07/2000	CW	07/09/2004												SB 72-C925 EMBODIED PREVIOUSLY. NEW BEARING INSTALLED
005-04-2002 SUPERSEDES 004-08-2000	2004-01-21	Rolls-Royce plc (RR) RB211-22B, RB211-524 & RB211-537 series. To prevent failure of the front engine mount due to cracks, that could result in loss of the engine	39-13435	MSB RB. 211-71-D437 R01 RT000	27/02/2004	CW	07/09/2004												New Spherical bushing installed
CAA 005-07-99 R0		Radial drive steady bearing inspection		RB211-72-C815 R04 RT000	N/A	CW	07/09/2004												SB 72-C925 EMBODIED PREVIOUSLY. NEW BEARING INSTALLED
CAA 005-10-92 R2		Fuel and control - Fuel flow governor -i		RB211-73-9363 R01	N/A	CAN				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	CANCELLED BY RR
CAA 005-05-99 R2 SUPERSEDED BY 2009-09-14		RADIAL DRIVE STEADY BEARING INSPECTION		72-C810R2	N/A	SUP				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	SUPERSEDED BY CAA 004-12-99
EASA 006-06-85 R0		HP COMPRESSOR WITHDRAWAL OF SPECIFIC HP COMPRESSOR ROTOR STAGE 1 TO 2 DISC ASSEMBLIES FROM SERVICE				N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	
EASA 006-08-83		ENGINE FUEL AND CONTROL DEDICATED GENERATOR STATOR			22/11/2012	N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	APPLICABLE TO RB211-535C ENGINES
CAA 007-01-97 R2 SUPERSEDED BY 98-07-07		INSPECTION LP FUEL TUBE FLEXIBLE SECTIONS				SUP				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	SUPERSEDED BY AD 007-10-97
CAA 007-07-90 R2		NEW LOWER TELEFLEX GEARBOX COVER BRACKET		RB211-76-9170 R01 RT000	N/A	CW													APPLIC TO ENGINES PRIOR TO SN 30751 (E4) AND (E4B) 76-9170 EMBODIED PREVIOUSLY DURING MANUFACTURE

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CAA 007-10-97 R0 SUPERSEDED BY 2014-0123	98-07-07	INSPECTION LP FUEL TUBE FLEXIBLE SECTIONS	39-10426	RB211-73-C297 R04 RT000 RB211-73-B987 R05 RT000	29/04/1998	SUP				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	SB CANCELLED AT REV 4 SUPERSEDED BY CAA AD 2014-0123
EASA 009-02-87 R0		HPC ENGINE LIFE MANAGEMENT POLICY AND REVISED OPERATING PROCEDURES TO REDUCE THE POSSIBILITY OF COMPRESSOR SURGE DURING TAKE-OFF				N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	APPLICABLE TO RB211-535C ENGINES
CAA 009-03-93 R2 SUPERSEDED BY 007-10-97		FUEL TUBE FERRUEL RETAINER INSPECTION		73-9664R4	N/A	SUP				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	SUPERSEDED BY CAA 007-10-97
CAA 012-03-85 R2		ENGINE FUEL & CONTROL- RE-RUN P1 AIR TUBE INCORPORATING A RELIEF VALVE AND WATER DRAIN		RB211-73-7951 R01	N/A	N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	APPLIES TO ESN PRIOR TO 30523
CAA 013-03-85 R0		ENGINE FUEL & CONTROL- FFG BLANKING OF P1 VENT HOLE IN THE ALTITUDE BIAS SENSOR		RB211-73-7905 R00 RT000	N/A	N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	APPLY TO ESN 30523 TO 30529
CAA 016-03-93 R2		LP FUEL TUBE RETAINING RINGS DELETE FROM HP PUMP END		RB211-73-9761 R0	N/A	N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	APPLIC TO ENGINES PRIOR TO SN 30893. 73-9761 SUPERSEDED BY 73-B048 SS BY 73-C297 SB 73-C297 EMBODIED PREVIOUSLY DURING MANUFACTURE
004-12-99 SUPERSEDES 006-05-99	2000-09-14	RB211-535 radial drive steady bearing	39-11723	RB211-72-C930 R04 RT000	11/07/2000	CW	07/09/2004			N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	SB 72-C925 EMBODIED
004-09-2000	2001-19-05	RADIAL STEADY BEARING CONTROL ON AIRCRAFT	39-12445	RB211-72-C925R2 RB211-72-C815R3	11/10/2001	CW	07/09/2004			N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	SB 72-C925 EMBODIED
003-12-99	2001-23-02 SUPERSEDED BY 2004-26-03	HP turbine Disc-Eddy current inspection. To prevent possible disc failure, which could result in an uncontained engine failure and damage to the airplane	39-12499	RB211-72-C817 R02	24/12/2001	SUP				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	CAA AD 003-12-99 CANCELLED AND SUPERSEDED BY CAA AD G-2004-0027. FAA 2001-23-02 CANCELLED AND SUPERSEDED BY FAA AD 2004-26-03
006-05-2001	2002-23-08 SUPERSEDED BY 2003-17-15	TO PREVENT MCHINING INDUCED CRACKING OF THE HP TURBINE DISC		RB211-72-D181	16/12/2002	SUP													SUPERSEDED BY 2003-17-15
	2003-10-03R2	Removal from service of certain HPT discs before reaching new established life. Applicable only for rb211-535e4-b-37 & rb211-535e4-b-75			20/06/2003	N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	HPT DISC PN UL39767 IS INSTALLED
	2003-12-15 C SUPERSEDED BY 2005-02-05	CRACKS IN LPC FAN BLADE ROOTS INSPECTION		RB211-72-C879	30/07/2003	SUP													SUPERSEDED BY AD 2005-02-05
	2003-12-15 F SUPERSEDED BY 2005-02-05 F	CRACKS IN LPC FAN BLADE ROOTS		RB211-72-C946	30/07/2003	SUP													SUPERSEDED BY AD 2005-02-05
006-05-2001	2003-17-15 SUPERSEDES 2002-23-08	LP TURBINE ROTOR- 2ND STAGE DISK REDUCED CYCLIC LIFE ENGINE MANAGEMENT AND ON WING INSPECTION	39-13290	RB211-72-D181 R04	02/10/2003	CW				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	LIFE CONTROL PER TLM MANUAL. NO ACTION REQUIRED BY AD
004-01-94	2004-01-20 R1 SUPERSEDED BY 2006-11-05	TO PREVENT CORROSION INDUCED UNCONTAINES DISC FAILURE RESULTING IN DAMAGE TO THE AIRPLANE			24/02/2004	SUP													SUPERSEDED BY AD 2006-11-05
005-04-2002	2004-01-21	Rolls-Royce plc (RR) RB211-22B, RB211-524 & RB211-537 series. To prevent failure of the front engine mount due to cracks, that could result in loss of the engine	39-13435	MSB RB. 211-71-D437 R01	27/02/2004	CW	9-jul.-04												SB 71-D437 EMBODIED
G-2003-0007	2004-19-04 A	ACTIONS TO PREVENT FAILURE OF CRITICAL LIFE-LIMITED ROTATING ENGINE PARTS			27/10/2004	OPEN							SV	SV	SV	SV	SV	SV	EACH SHOP VISIT WHEN THE COMPONENT IS COMPLETELY DISASSEMBLED TO PIECE-PART LEVEL
G-2003-0007	2004-19-04 B	FAILURE OF CRITICAL LIFE-LIMITED PARTS			27/10/2004	CW													AMP HAS BEEN REVISED AT 2005
G-2004-0027	2004-26-03 R1 SUPERSEDED BY 2006-17-12	PREVENT POSSIBLE DISC FAILURE		RB211-72-AE651	13/01/2005	SUP													SUPERSEDED BY AD 2006-17-12
	2005-02-05 C SUPERSEDED BY 2008-22-24	CRACKS IN LPC FAN BLADE ROOTS			11/02/2005	SUP													SUPERSEDED BY 2008-22-24
	2005-02-05 F SUPERSEDED BY 2003-12-15 F	CRACKS IN LPC FAN BLADE ROOTS		RB211-72-C946	11/02/2005	N/A													BLADES TO SB72-C946 STANDARD THAT DO NOT REQUIRED INSPECTION
2006-0182	2008-20-04	HPT Discs recently inspected in accordance with the Engine Manual have exhibited cracks in the disc rim. The discs have failed to meet the inspection acceptance criteria		SB RB211-72-AE969 R0	10.07.2006	OPEN							SV	SV	SV	SV	S.V.	S.V.	NEXT SHOP VISIT WHEN THE PARTS ARE COMPLETELY DISASSEMBLED TO PIECE PART LEVEL
004-01-94	2006-11-05 R2	HPC STG 3 DISC REVISED LIVES FOR REWORK IAW SB 72-9434	39-13434	RB211-72-9434 R04 RB211-72-9661 R04	21/02/2008	OPEN							21/10/2027	15000	2416		9723		PN:UL24738
G-2004-0027 2004-11146	2006-17-12 SUPERSEDES 2004-26-03	HP TURBINE DISC COOLING AIR HOLE CRACKING	39-13915	SB RB211-72-AE651 R0	13/01/2005	N/A													HP DISC PN UL39767 INSTALLED

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G-2005-0028R1	2007-14-07	HPC STAGE 1 AND 2 ROTOR ENHANCED INSPECTION OF THE INTERSTAGE WELD	39-15129	ASB RB.211-72-AE359 R01	16/08/2007	OPEN							SV	SV	SV	SV	S.V.	S.V.	NEXT SHOP VISIT WHEN THE PARTS ARE COMPLETELY DISASSEMBLED TO PIECE PART LEVEL
2008-0046		ENGINE-LOW PRESSURE (LP) TURBINE DISCS		SB RB211-72-AF923 R0	27/03/2008	CAN				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	CANCELLED JULY 9 2009
AD 2008-0176		HIGH PRESSURE TURBINE DISC-INSPECTION		RR NM5B 72-AF996 R01	02/10/2008	N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	HPT DISC NOT AFFECTED DUE TO AFFECTED SN OF HPT DISCS ARE NOT INSTALLED
2006-0182	2008-20-04	HPT Discs recently inspected in accordance with the Engine Manual have exhibited cracks in the disc rim. The discs have failed to meet the inspection acceptance criteria		SB RB211-72-AE969 R00 RT001	03/11/2008	OPEN							SV	SV	SV	SV	S.V.	S.V.	NEXT SHOP VISIT WHEN THE PARTS ARE COMPLETELY DISASSEMBLED TO PIECE PART LEVEL
2016-0202R1	2008-22-24 SUPERSEDES 2005-02-05 C	FAN BLADE ROOTS ULTRASONIC INSPECTION		RB211-72-C879 R9	25/09/2018	CW						1150 1400			1150				ALL FAN BLADES ARE PN FW
AD 2009-0187R2 SUPERSEDES 2009-0187R1		To prevent cracking on the head sections on the front combustion liners that could result in an unsafe condition in which hot gas breakout with subsequent downstream component release leads to uncontained high energy debris. It mandates initial inspection of the head section of the Front Combustion Liner (FCL) and possible repetitive inspections or replacement of the FCL		SB RB.211-72-AG073	20/05/2011	N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	ENGINE PRE SB 72-C230
2009-0243R2 SUPERSEDES 2009-0243R1		ENGINE -FRONT COMBUSTION LINER INNER WALL INSPECTION		RB211-72-AF458R4 RB211-72-D133 RB211-72-E902	02/12/2009	N/A				N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	R2 OF AD IS ISSUED TO REMOVE FROM THE APPLICABILITY PARAGRAPH OF AD ALL MODELS OF THE RB211-535 SERIES ENGINES
	2001-11-08 SUPERSEDED BY 2012-02-11	LPT STAGE 1,2,3 INSPECTION			30/06/2011	SUP													
2009-0244	2012-02-11 SUPERSEDES 2011-11-08	LPT STAGE 1 DISC INSPECTION LPT STAGE 2 DISC INSPECTION LPT STAGE 3 DISC INSPECTION	39-16934	72-AG272	09/12/2009	OPEN							S.V	S.V	S.V	S.V	S.V	S.V	inspection to be performed at every sv if more than 1500cyc have been accumulated since previous inspection
008-07-85	2009-20-14	COMB OUTER CASE ON WING CRACK INSPECTION AT THE HP 6 OFF- TAKE SOLE PLATE PRE SB 72-8045		72-7775R3	17/11/2009	N/A							N/A	N/A	N/A	N/A	N/A	N/A	SB 72-8045 EMBODIED PREVIOUSLY
2012-0060	2012-25-09	IP TURBINE DISC- REMOVAL DUE TO STEEL INCLUSION CONCERNS	N/A	72-AG493	02/05/2012	N/A							N/A	N/A	N/A	N/A	N/A	N/A	IPT DISC INSTALLED NOT AFFECTED BY SB 72-AG493R2
2012-0215R1 SUPERSEDES 2012-0215	2013-16-23	FRONT COMBUSTION LINER METERING PANEL INSPECTION/REPLACEMENT	39-17561	72-AG046 72-AF572R2	08/11/2013	CW													SB 72-AF572R02 & 72-AG046R04 WAS PREVIOUSLY EMBODIED
2012-0265	2013-19-17	IMPLEMENTATION OF TWO NEW FLIGHT PROFILES FOR RB211-535E4-B-37 ENGINES		72-AG875	01/01/2013	N/A							N/A	N/A	N/A	N/A	N/A	N/A	FOR RB211-535E4-B-37 ENGINES ONLY
2012-0266	2013-19-18	LPT STAGE 2 DISCS LIFE LIMITS		RB211-72-AH029	01/01/2013	CW													AMP LIFE LIMIT SECTION WAS REVISED AND LPT ST2 DISCS HAVE CORRECT LIFE LIMIT AS PER SB 72-AH029
2013-0165	2013-23-17	HP TURBINE DISCS- REMOVAL DUE TO STEEL CONTAMINATION CONCERNS		72-AH215	01/08/2013	N/A							N/A	N/A	N/A	N/A	N/A	N/A	HPT DISC NOT AFFECTED
2014-0123 SUPERSEDED BY 2017-0006	2014-24-08 SUPERSEDES 98-07-07	INTRODUCTION OF A RIGID LP FUEL FILTER TO HP FUEL PUMP TUBE ASSY AND ADDITIONAL LINE BLOCK SUPPORTS CLAMPS	39-10426	73-H131	06/01/2015	CW	18-oct.-16												SB 73-H131 EMBODIED AT SV IN AMECO, DATED 2016-10-18
AD 2014-0249R1		An engineering analysis, carried out by RR, of the lives of critical parts of the RB211-535E4-37 engine, has resulted in reduced cyclic life limits for certain high pressure (HP) turbine discs. The reduced limits are published in the RR RB211-535E4-37 Time Limits Manual (TLM): 05-10-01-800-000, current Revision dated July 2014.	N/A	RR RB211-535E4-37 TLM 05-10-01-800-000, Revision dated 01 July 2014. 72-G188	03/12/2014	CW							LLP	LLP	LLP	LLP	LLP	LLP	SEE ENGINE LLP STATUS
AD 2015-0058 SUPERSEDED BY 2015-0103		A review of operational flight data has revealed that some RB211-535 engines may have been operated beyond the currently valid datum flight profile (FP) published in the applicable Aircraft Maintenance Manuals. The purpose of the datum FPs are to establish the operational limits (life limits) within which the corresponding critical parts are allowed to remain installed.		Rolls Royce plc. WWW11174/2, dated 22 January 2015.	09/04/2015	SUP													Superseded by AD 2015-0103 R1
AD 2015-0103 R01 SUPERSEDED BY 2015-0148		A review of operational flight data has revealed that some RB211-535 engines may have been operated beyond the currently valid datum flight profile (FP) published in the applicable Aircraft Maintenance Manuals. The purpose of the datum FPs are to establish the operational limits (life limits) within which the corresponding critical parts are allowed to remain installed.	N/A	72-AH972 R2	12/06/2015	SUP													Superseded by AD 2015-0148
AD 2015-0148 SUPERSEDES 2015-0103		A review of operational flight data has revealed that some RB211-535 engines may have been operated beyond the currently valid datum flight profile (FP) published in the applicable Aircraft Maintenance Manuals. The purpose of the datum FPs are to establish the operational limits (life limits) within which the corresponding critical parts are allowed to remain installed.	N/A	72-AH972R3	06/08/2015	N/A							N/A	N/A	N/A	N/A	N/A	N/A	PREVIOUSLY COMPLIED IAW SB 72-AH972, N/A PER FLIGHT PROFILE AND EFFECTIVE LLP SN

JH

AIRWORTHINESS DIRECTIVES STATUS REPORT

FAA BW 2021-04 and EASA BW 2021-04

Fecha / Date:	10/03/2021
Horas de Vuelo / FH:	72701:00
Ciclos de Vuelo / FC:	24123

EASA/CAA	FAA	SUBJECT	AMEND	SB	ED	STAT	LAST ACC. DATE	LAST ACC. FH	LAST ACC. FC	INTERVAL			NEXT DUE			REMAINING			REMARKS
										DAYS	FH	FC	DATE	FH	FC	DAYS	FH	FC	
	2015-17-21	HPT DISK REMOVAL		72-G188	02/12/2015	CW													LIFE CONTROL PER TLM MANUAL. NO ACTION REQUIRED BY AD. LIFE LIMIT WAS PREVIOUSLY REDUCED IAW SB 72-G188 REQUIREMENTS
2016-0031		ENGINE HEALTH CHECK AND REVISED OPERATING PROCEDURES-CRITICAL PART IDENTIFICATION AND REPLACEMENT			08/03/2016	N/A							N/A	N/A	N/A	N/A	N/A	N/A	N/A PER ENGINE TYPE
	2016-03-04	FLIGHT PROFILE G			15/03/2016	N/A							N/A	N/A	N/A	N/A	N/A	N/A	N/A PER FLIGHT PROFILE AND EFFECTIVE LLP SN
2017-0006 SUPERSEDES 2014-0123	2017-24-08	FUEL TUBE AASY (LP FUEL TUBES/HP FUEL TUBES/FLANGED ADAPTORS) REPLACEMENT		RB211-73-H131 RB 73-G2305	24/01/2017	CW	25/10/2010 17/11/2016												SB 73-G230 WAS EMBODIED AT SV DATED 25.10.2010. SB RB211-73-H131R1 WAS INCORPORATED DURING SV AT AMECO DATED 27.11.2016
2018-0202R1	2008-22-24 SUPERSEDES 2005-02-05 2020-08-03 supersedes 2008-22-24 & 2018-0202R1	FAN BLADE ROOTS ULTRASONIC INSPECTION		RB211-72-C879 R9	25/09/2018	N/A	10/09/2015	48762	26031			1150 1400				N/A	N/A	N/A	ALL FAN BLADES ARE PN FW. SUPERSEDED BY AD 2020-08-03
		FAN BLADE ROOTS ULTRASONIC INSPECTION	39-21109	RB211-72-C879 R9	18/05/2020	N/A	04/06/2019	51757	27147			1150 1400				N/A	N/A	N/A	ALL FAN BLADES ARE PN FW.

 Alice Wang
MM Planning Manager



PRIVILEGE
BOEING 767-300ER-777 style
**DIRECCIÓN TÉCNICA
MANTENIMIENTO**



SECTION 8

INVENTORY LIST



A J Walter Leasing Limited, Exchange House, 54-58 Athol Street, Douglas, IM1 1JD, Isle of Man, T: +44 7552 812215 E: engines@ajw-leasing.com W: www.ajw-aviation.com

ATA	Part Number	S/N	Descripción P/N	INSTALLATION DATE	TT AT INSTALLATION	CYC AT INSTALLATION
77	3301KGA/MS/1	981963	N3 TACHOMETER GENERATOR	21-sep.-19	71848	23848
74	YA211-25	18027473	PLUG, HE IGNITER	NOTE 1		
75	ETPU100-03C	BX31099	TPU			
75	789830-1	98077656	BVCU			
75	312N5321-7	5B020015	DUCT ASSY HP6	NOTE 1		
75	AC67760	RG207	SOLENOID IP BLEED VALVE	NOTE 1		
75	AC67760	PJ188	SOLENOID IP BLEED VALVE	NOTE 1		
75	AC67926	EGL266	VALVE ASSY HANDLING HP3 BLEED LH	NOTE 1		
75	AC67926	PK053	VALVE ASSY HANDLING HP3 BLEED RH	NOTE 1		
75	5496-7-49	1684	THEMCPL, ASSY SONIC T2	NOTE 1		
76	44263-3	302	CABLE ASSY CONTROL FUEL CUT-OFF	NOTE 1		
76	ESV010	2476385	VALVE, SHUT OFF-EMERGENCY	NOTE 1		
77	684-4-33	82131	THERMOCUPLE T7	NOTE 1		
77	684-4-33	76809	THERMOCUPLE T7	NOTE 1		
77	684-4-33	76921	THERMOCUPLE T7	NOTE 1		
77	684-4-33	76837	THERMOCUPLE T7	NOTE 1		
77	684-4-33	66342	THERMOCUPLE T7	NOTE 1		
77	684-4-33	76918	THERMOCUPLE T7	NOTE 1		
77	684-4-33	82137	THERMOCUPLE T7	NOTE 1		
77	684-4-33	76635	THERMOCUPLE T7	NOTE 1		
77	684-4-33	82164	THERMOCUPLE T7	NOTE 1		
77	684-4-33	77154	THERMOCUPLE T7	NOTE 1		
77	684-4-33	76429	THERMOCUPLE T7	NOTE 1		
77	684-4-33	82128	THERMOCUPLE T7	NOTE 1		
77	684-4-33	82181	THERMOCUPLE T7	NOTE 1		
77	684-4-33	84447	THERMOCUPLE T7	NOTE 1		
77	684-4-33	84449	THERMOCUPLE T7	NOTE 1		
77	684-4-33	82132	THERMOCUPLE T7	NOTE 1		
77	684-4-33	82129	THERMOCUPLE T7	NOTE 1		
79	37757-3009	SV0507	VALVE OIL BY-PASS	NOTE 1		
79	8TJ99GBU2	1680T	TRNSMTTR,ASSY OIL QUANTITY	NOTE 1		
80	774987-1	200104087	VALVE, CONTROL PNEUMATIC STARTER	NOTE 1		
24	727792N	AA013	IDG	NOTE 1		
26	UL26042	M721010	SENSOR FIRE RH	NOTE 1		
26	UL26048	M351670	SENSOR FIRE LH	NOTE 1		
29	350880-6	MX220989B	PUMP, HYDRAULIC	NOTE 1		
30	7646B000	757622-1	VALVE, PRESSURE REGULATING	NOTE 1		
72	QB0163	PY5-135	FILTER AIR CENTRISEP TO BLOWN SEAL	NOTE 1		
73	BPU200MK2	B460	PUMP, ASSY FUEL LP	NOTE 1		
73	GP202MK1	B412	PUMP, ASSY FUEL HP	NOTE 1		
73	P39-401REVF	9050344	VALVE,DRAINS TANK FUEL RET	NOTE 1		
73	FFG022AC	2245338	GOVERNOR, FLOW FUEL	NOTE 1		
74	YA211-25	18027467	PLUG, HE IGNITER	NOTE 1		

Note 1: installed according AMECO engine accessory report dated 30jan2019

Alice wang

Planning Manager

